

Leveraging physics information in neural networks for fluid flow problems

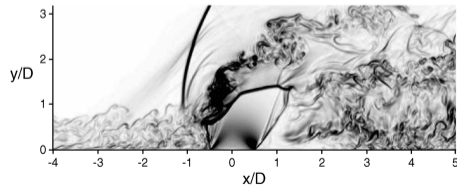
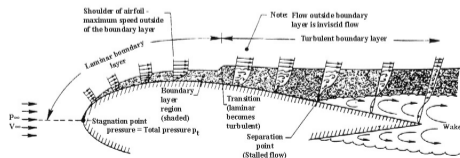
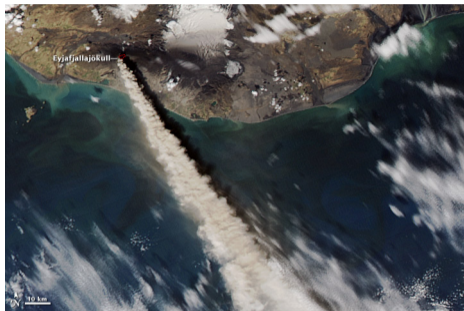
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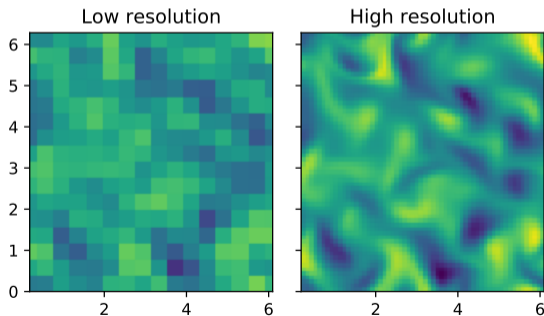
March 17, 2021

Turbulence



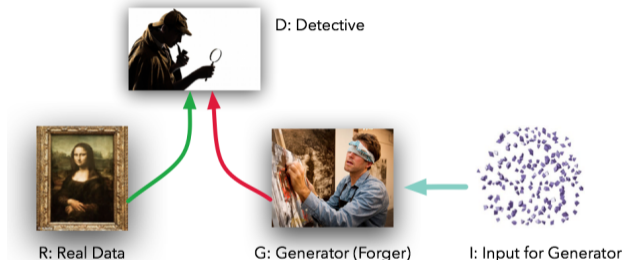
Turbulence Enrichment

Given a low-resolution turbulent flow field (LES), recover the high-resolution field (DNS) in a pointwise sense



Generative Adversarial Networks

- A class of generative models introduced by Goodfellow et. al.¹



- Mainly used for generating photorealistic images
- Also used previously for scientific datasets²³

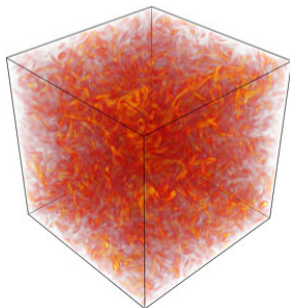
¹Ian Goodfellow et al. "Generative adversarial nets". In: *Advances in neural information processing systems*. 2014, pp. 2672–2680.

²Mustafa Mustafa et al. "Creating Virtual Universes Using Generative Adversarial Networks". In: *arXiv preprint arXiv:1706.02390* (2017).

³Shing Chan and Ahmed H Elsheikh. "Parametrization and Generation of Geological Models with Generative Adversarial Networks". In: *arXiv preprint arXiv:1708.01810* (2017).

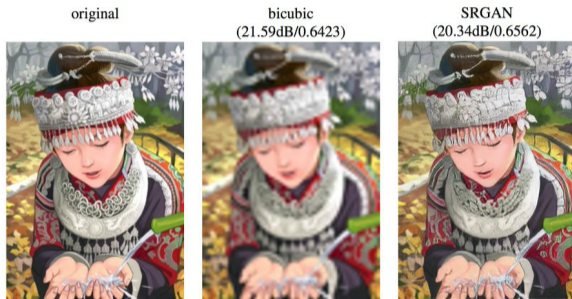
Data

- 4 3D fields: 3 components of velocity and pressure
- Each field is on a $64 \times 64 \times 64$ grid
- Low resolution data: filtered and downsampled to $16 \times 16 \times 16$
- Homogeneous Isotropic Turbulence (HIT) that is stationary in time
- A fairly low Reynolds number case to keep computational costs low



SRGAN

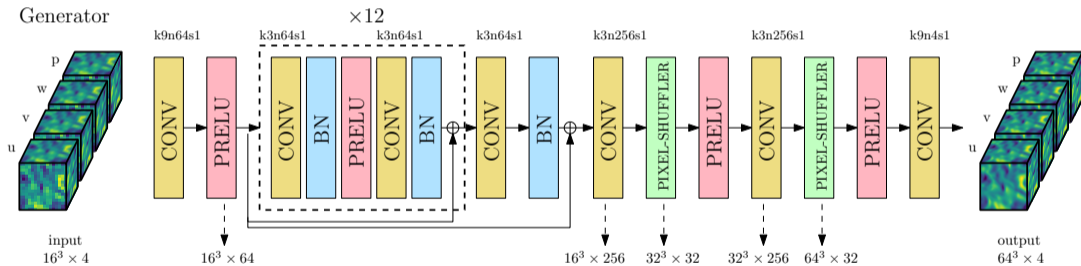
- SRGAN: super-resolution for images⁴



- Residual network with convolutional layers
- Two upsampling layers
- Model architecture used in this work is inspired by SRGAN

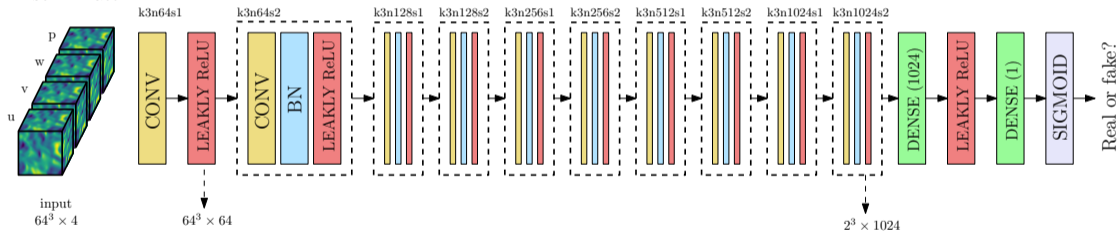
⁴Christian Ledig et al. "Photo-realistic single image super-resolution using a generative adversarial network". In: *arXiv preprint (2016)*.

Generator architecture



Discriminator architecture

Discriminator



Physics informed learning



- The generated results need to be physical
- Respect conservation laws governing the system

$$\begin{aligned}\nabla \cdot \mathbf{u} &= 0, \\ -\nabla^2 p &= \nabla \mathbf{u} : \nabla \mathbf{u}^T\end{aligned}$$

- Can do so by penalizing the generator using the residual of the equations above

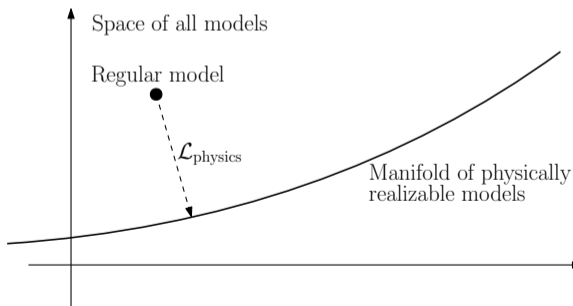
$$\begin{aligned}\mathcal{L}_{\text{continuity}} &= \int_{\Omega} (\nabla \cdot \mathbf{u})^2 \, d\Omega, \\ \mathcal{L}_{\text{pressure}} &= \int_{\Omega} \left(\nabla^2 p + \nabla \mathbf{u} : \nabla \mathbf{u}^T \right)^2 \, d\Omega\end{aligned}$$

Physics loss

- Add a physics loss to the generator training

$$\mathcal{L}_{\text{physics}} = (1 - \lambda_C) \mathcal{L}_{\text{pressure}} + \lambda_C \mathcal{L}_{\text{continuity}}$$

- Enforces better compatibility with physics
- Acts as a regularizer for the model



Loss function

- The loss function \mathcal{L}_{GAN} is given by

$$\mathcal{L}_{GAN} = (1 - \lambda_A) \mathcal{L}_{resnet} + \lambda_A \mathcal{L}_{adversarial}$$

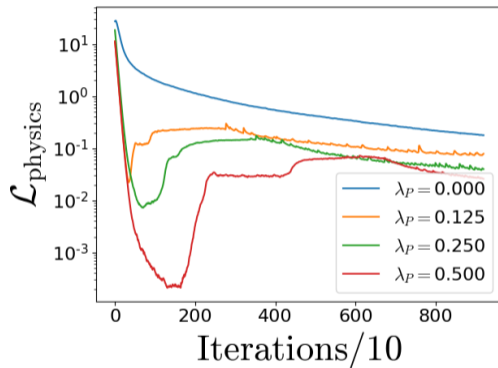
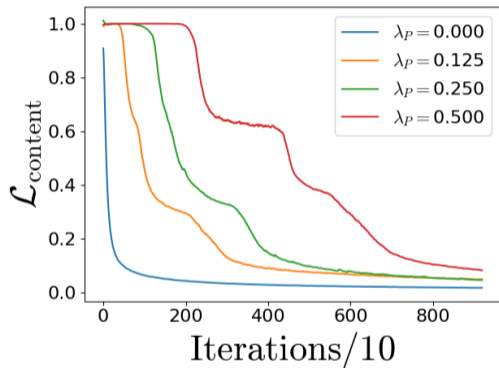
$$\mathcal{L}_{resnet} = (1 - \lambda_P) \mathcal{L}_{content} + \lambda_P \mathcal{L}_{physics}$$

$$\mathcal{L}_{content} = (1 - \lambda_E) \mathcal{L}_{MSE} + \lambda_E \mathcal{L}_{enstrophy}$$

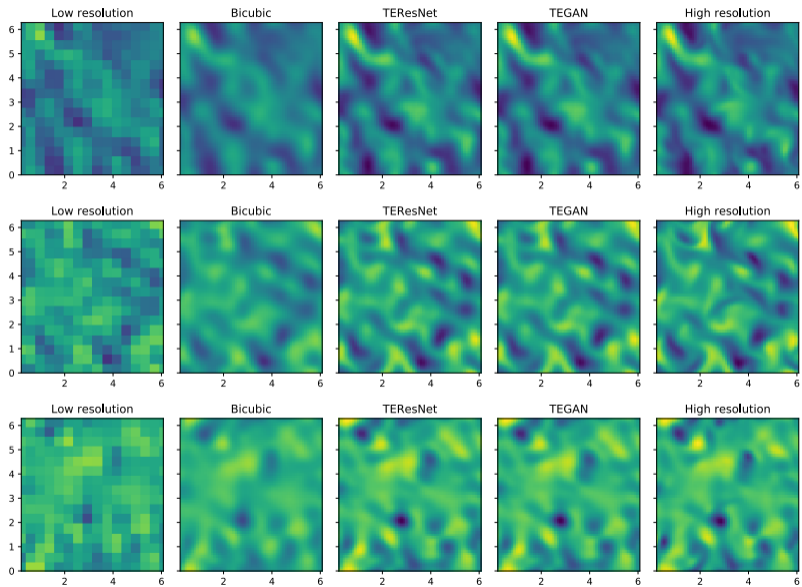
$$\mathcal{L}_{physics} = (1 - \lambda_C) \mathcal{L}_{pressure} + \lambda_C \mathcal{L}_{continuity}$$

- MSE and enstrophy sensitize the model to large and small scale features respectively
- Four hyperparameters to tune: λ_A , λ_P , λ_E and λ_C

Impact of physics loss on training



- Increasing λ_P improves the physics loss by an order of magnitude but compromises content loss
- Higher λ_P creates a strong local minimum at the trivial solution
- Choose $\lambda_P = 0.125$ as a compromise



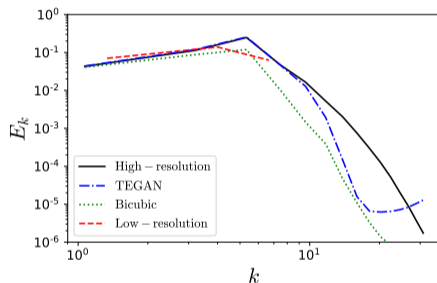
Evaluation

	$\mathcal{L}_{\text{content}}$		$\mathcal{L}_{\text{physics}}$	
	Dev	Test	Dev	Test
TEResNet	0.049	0.050	0.078	0.085
TEGAN	0.047	0.047	0.070	0.072
% Difference	4.1	6.0	10.3	15.2

- TEGAN has consistently lower content and physics losses
- TEGAN also generalizes to the dev set better

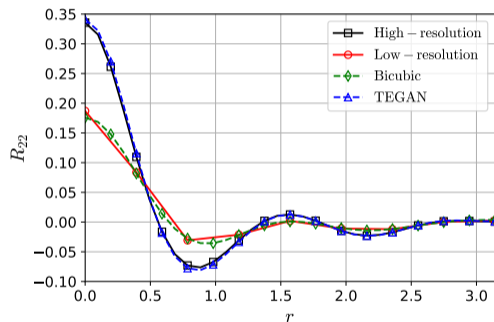
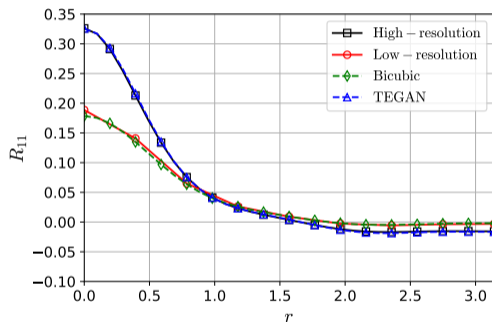
Energy spectra

- Energy spectra of the velocity field is a fundamental statistical quantity in turbulence



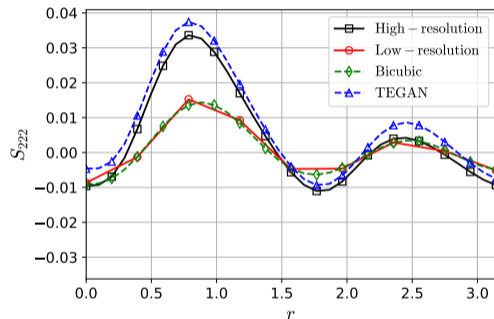
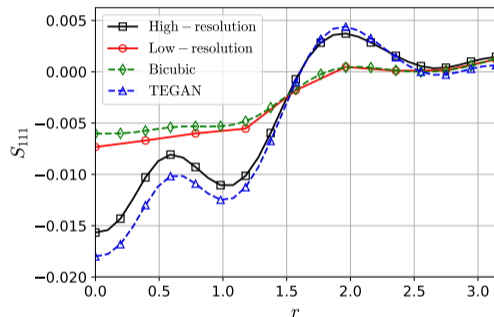
- The low resolution data is very coarse
- TEGAN is able to recover the spectrum very well in the intermediate wavenumbers
- Not as effective at capturing the finest dissipative scales

Second order two-point correlations



- $R_{11}(r = 0)$ and $R_{22}(r = 0)$ are the variances of the longitudinal and transverse velocity components respectively
- TEGAN captures both the longitudinal and transverse correlations virtually perfectly

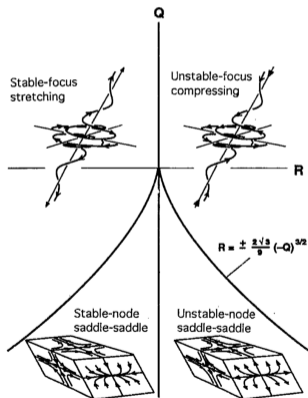
Third order two-point correlations



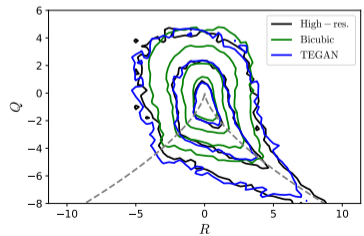
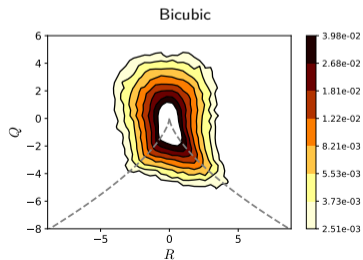
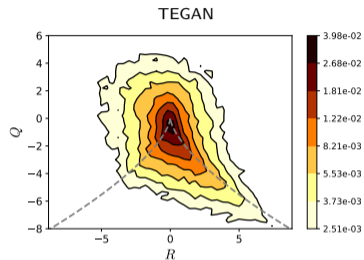
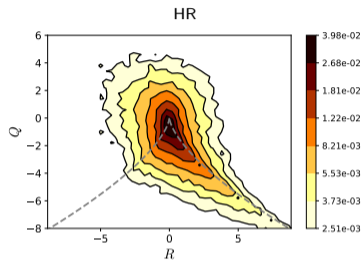
- TEGAN captures the qualitative structure of third order correlations
- $\approx 15\%$ overprediction of the longitudinal correlations
- Third order correlations are harder for models to capture

QR diagram

- The velocity gradient tensor gives a picture of the local flow structure
- Represent using the second and third invariants of the tensor: Q and R



QR diagram



Acknowledgements



- Man-Long Wong
- Raunak Borker
- Sravya Nimmagadda
- Sanjiva Lele



Physics-Informed Neural Networks (PINNs)

<http://developer.nvidia.com/simnet>

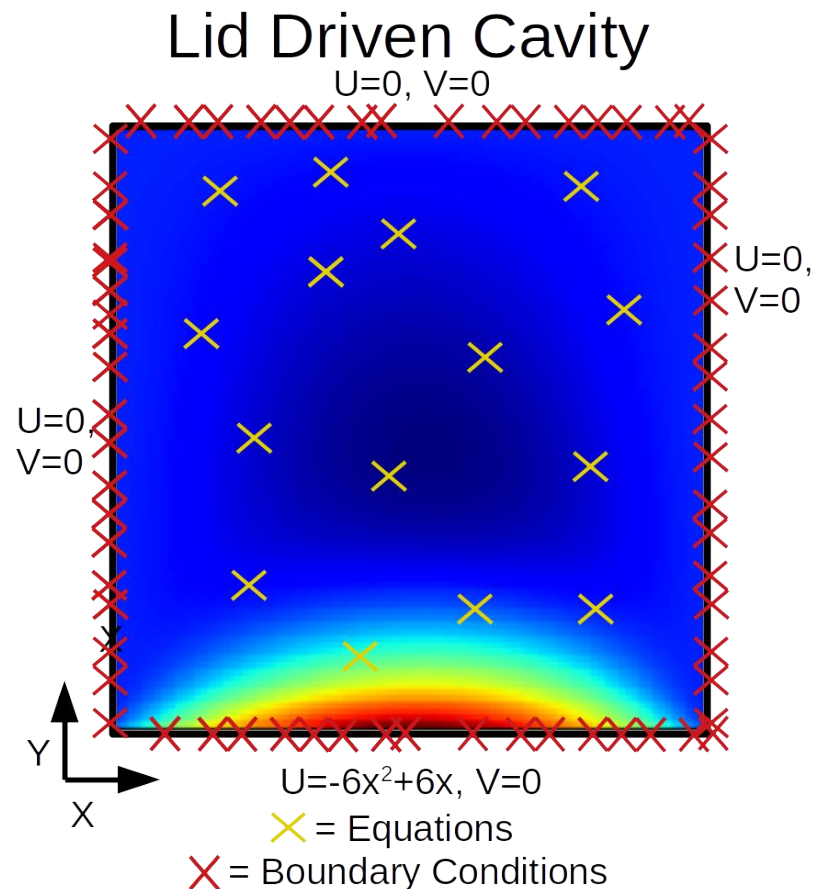
SOLVING PDEs WITH PINNs

Neural network approximates solution to partial differential equation.

$$u_{net}(x, y) \rightarrow (u, v, p)$$

Minimize loss from boundary conditions and equations.

$$0 = \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y}$$
$$0 = u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} + \frac{\partial p}{\partial x} - \nu \left(\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} \right)$$
$$0 = u \frac{\partial v}{\partial x} + v \frac{\partial v}{\partial y} + \frac{\partial p}{\partial y} - \nu \left(\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} \right)$$



NEURAL NETWORK SOLVER METHODOLOGY

The idea is to use a neural network to approximate the solution to given differential equation and boundary conditions.

Example Problem,

$$\mathbf{P} : \begin{cases} \frac{\delta^2 u}{\delta x^2}(x) = f(x), \\ u(0) = u(1) = 0, \end{cases} \quad (1)$$

Construct a deep multi-layer perception $u_{net}(x) \rightarrow u$. $x \in \mathbb{R}$. Assume that $u_{net} \in C^\infty$. This means using activation functions like *tanh*, *swish*, *sin*, *sigmoid*... [1]

NEURAL NETWORK SOLVER METHODOLOGY

Construct a Loss function to train $u_{net}(x)$. We can compute $\frac{\delta^2 u_{net}}{\delta x^2}(x)$ using automatic differentiation.

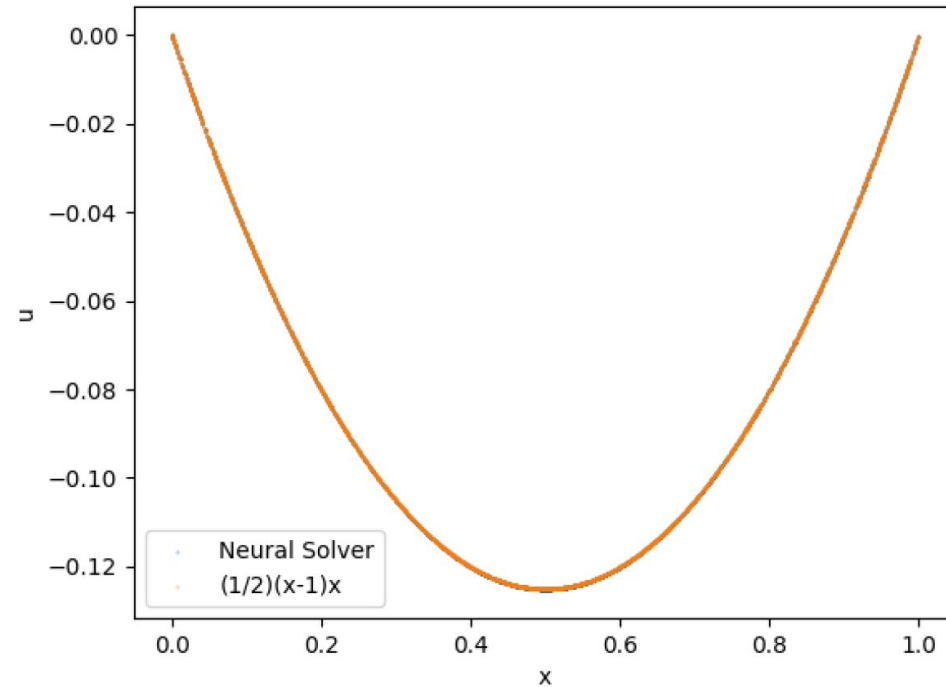
$$L_{BC} = u_{net}(0)^2 + u_{net}(1)^2 \quad (2)$$

$$L_{residual} = \frac{1}{N} \sum_{i=0}^N \left(\frac{\delta^2 u_{net}}{\delta x^2}(x_i) - f(x_i) \right)^2; x_i \in (0, 1) \quad (3)$$

$$L = L_{BC} + L_{residual} \quad (4)$$

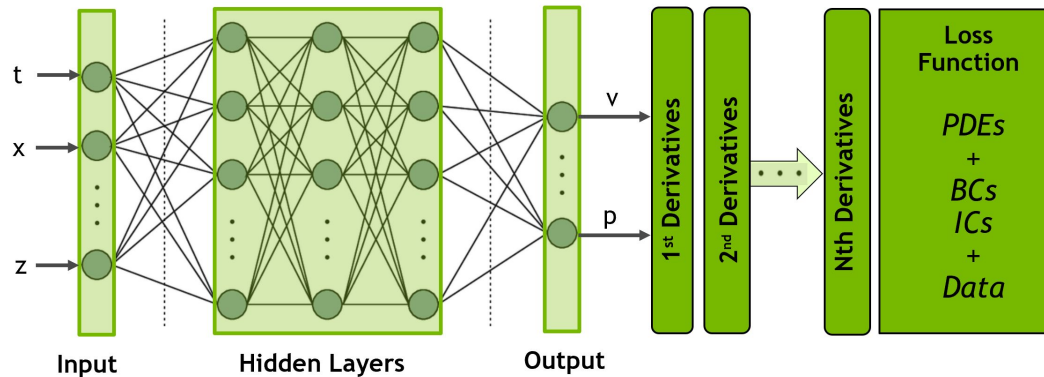
NEURAL NETWORK SOLVER METHODOLOGY

$$f(x) = 1$$



Physics Informed Neural Network Architectures

- Fully Connected (FC)



- Fourier Features (FN) - Axis, Partial, Full or Random Spectrum
- Sinusoidal Representation (SiReNs)
- Modified Fourier Features (mFN)
- Deep Galerkin Method (DGM)
- Modified Highway Networks

INVERSE PROBLEM

Inverse problems start with effects and then calculate the causes. Suppose we are given the solution $u_{true}(x)$ at 100 random points between 0 and 1 and we want to determine the $f(x)$ that is causing it.

$$L_{residual} \approx \left(\int_0^1 dx \right) \frac{1}{N} \sum_{i=0}^N \left(\frac{\delta^2 u_{net}}{\delta x^2}(x_i) - f_{net}(x_i) \right)^2; x_i \in (0, 1) \quad (10)$$

$$L_{data} = \frac{1}{100} \sum_{i=0}^{100} \left(u_{net}(x_i) - u_{true}(x_i) \right)^2 \quad (11)$$

INVERSE PROBLEM

For $u_{true}(x) = \frac{1}{48}(8x(-1 + x^2) - (3\sin(4\pi x))/\pi^2)$ the solution for $f(x)$ is $x + \sin(4\pi x)$.

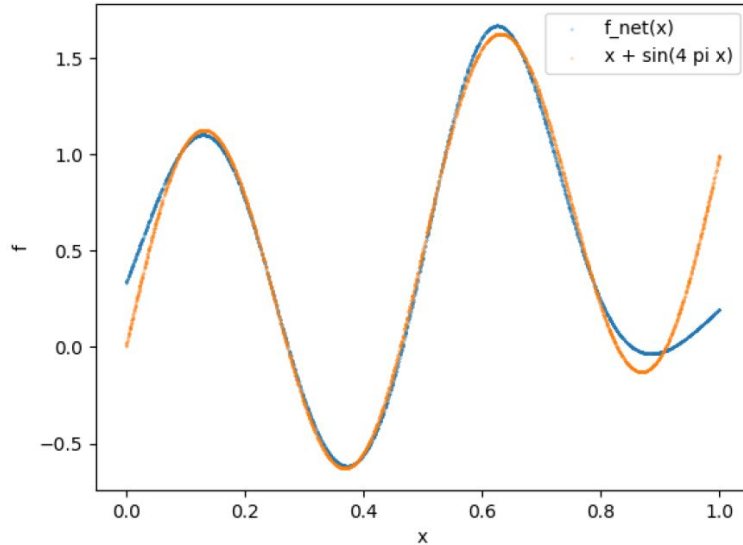


Figure: Comparison of true solution for $f(x)$ and the function inverted out.

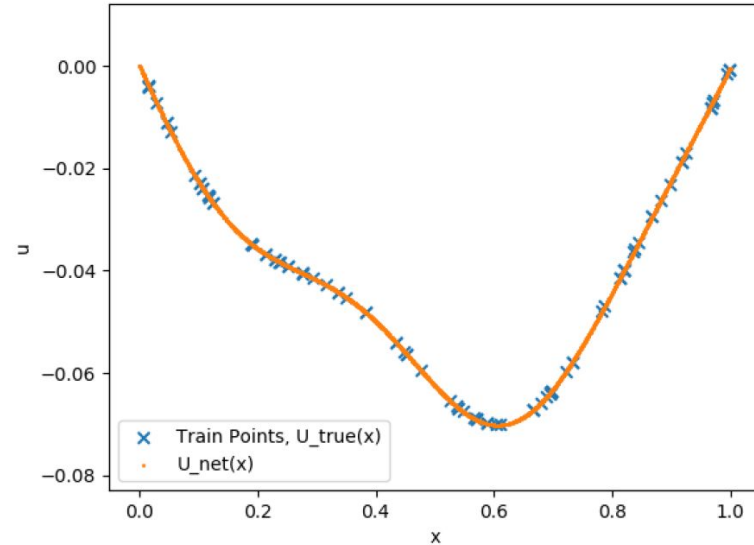


Figure: Comparison of $u_{net}(x)$ and train points from u_{true} .

SOLVING PARAMETERIZED PDES

We can solve parameterized geometries [2]. Suppose we want to know how the solution changes as we move the position on the boundary condition $u(1) = 0$.

$$\mathbf{P} : \begin{cases} \frac{\delta^2 u}{\delta x^2}(x) = f(x), \\ u(0) = u(l) = 0, l \in [1, 2] \end{cases} \quad (7)$$

Now train a network $u_{net}(x, l)$ on the losses,

$$L_{residual} \approx \left(\int_1^2 \int_0^l dx dl \right) \frac{1}{N} \sum_{i=0}^N \left(\frac{\delta^2 u_{net}}{\delta x^2}(x_i, l_i) - f(x_i) \right)^2 \quad (8)$$

$$L_{BC} \approx \left(\int_1^2 dl \right) \frac{1}{N} \sum_{i=0}^N \left(u_{net}(0, l_i) \right)^2 + \left(u_{net}(l_i, l_i) \right)^2 \quad (9)$$

SOLVING PARAMETERIZED PDES

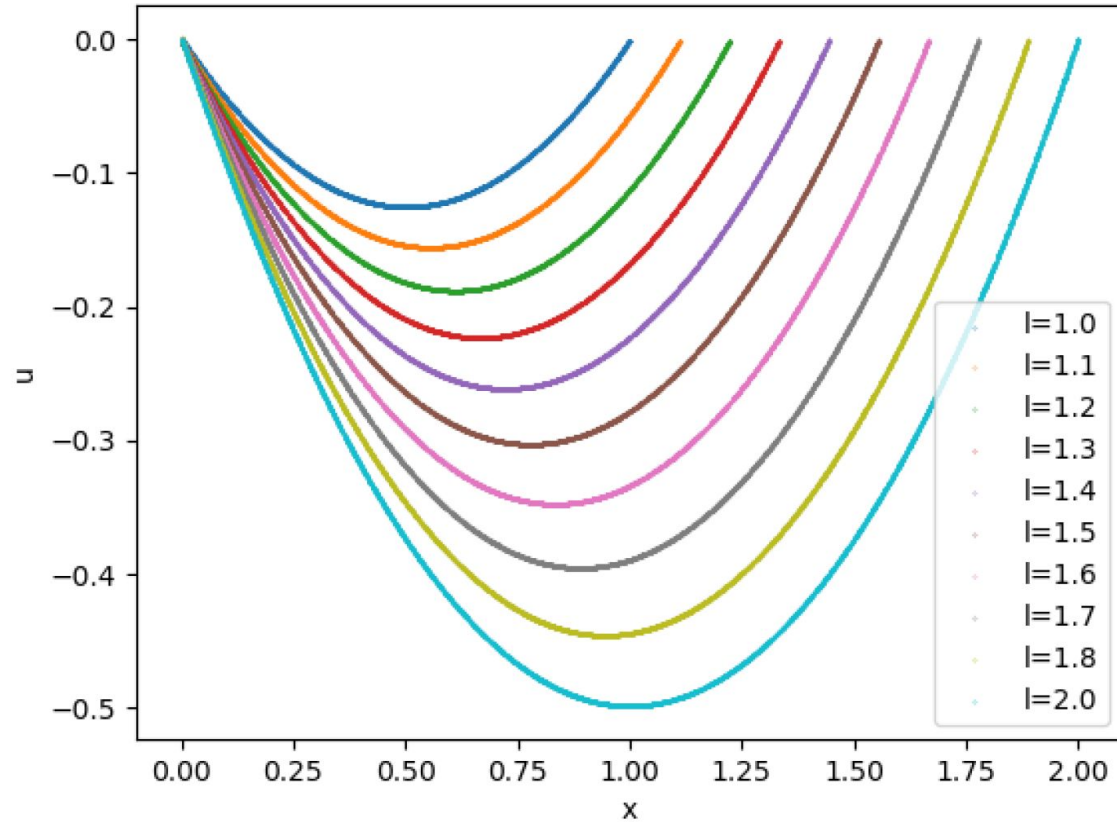
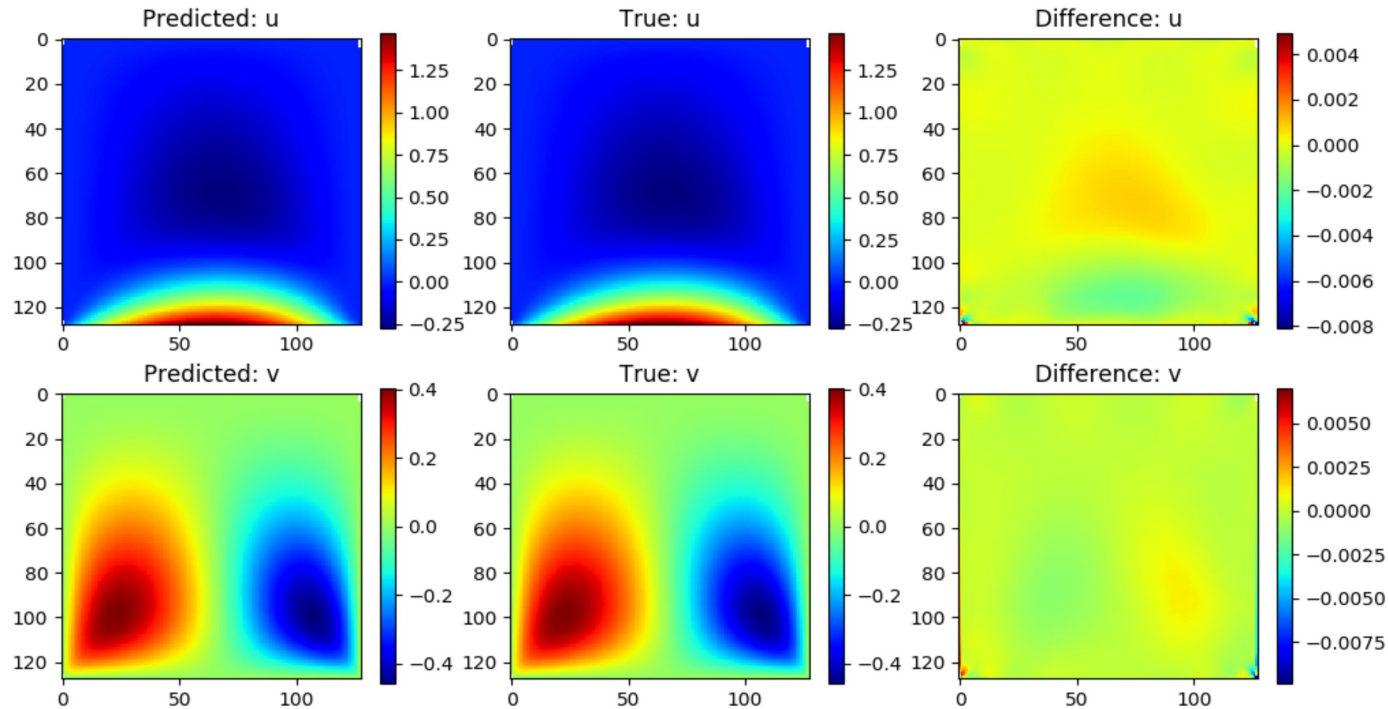


Figure: SimNet solving parameterized differential equation problem.

LID DRIVEN CAVITY

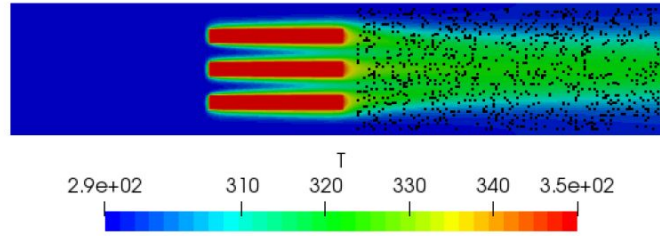
SimNet versus OpenFOAM



U velocity difference = 0.2%
V velocity difference = 0.4%

INVERSE PROBLEM APPLICATION

Finding Unknown Coefficients of a PDE: Heat Sink



Fluid Heat Convection:

$$0 = \nabla \cdot (D_{fluid} \nabla \theta_{fluid}) - \nabla \cdot (U \theta_{fluid}) \quad D_{fluid} = \frac{k_{fluid}}{\rho_{fluid} c_{pfluid}}$$

Solid Heat Conduction:

$$0 = \nabla \cdot (k_{solid} \nabla \theta_{solid}) \quad D_{solid} = \frac{k_{solid}}{\rho_{solid} c_{psolid}}$$

$$\theta_{solid} = \theta_{fluid}$$

Interface Conditions:

$$k_{solid} (N \cdot \nabla \theta_{solid}) = k_{fluid} (N \cdot \nabla \theta_{fluid})$$

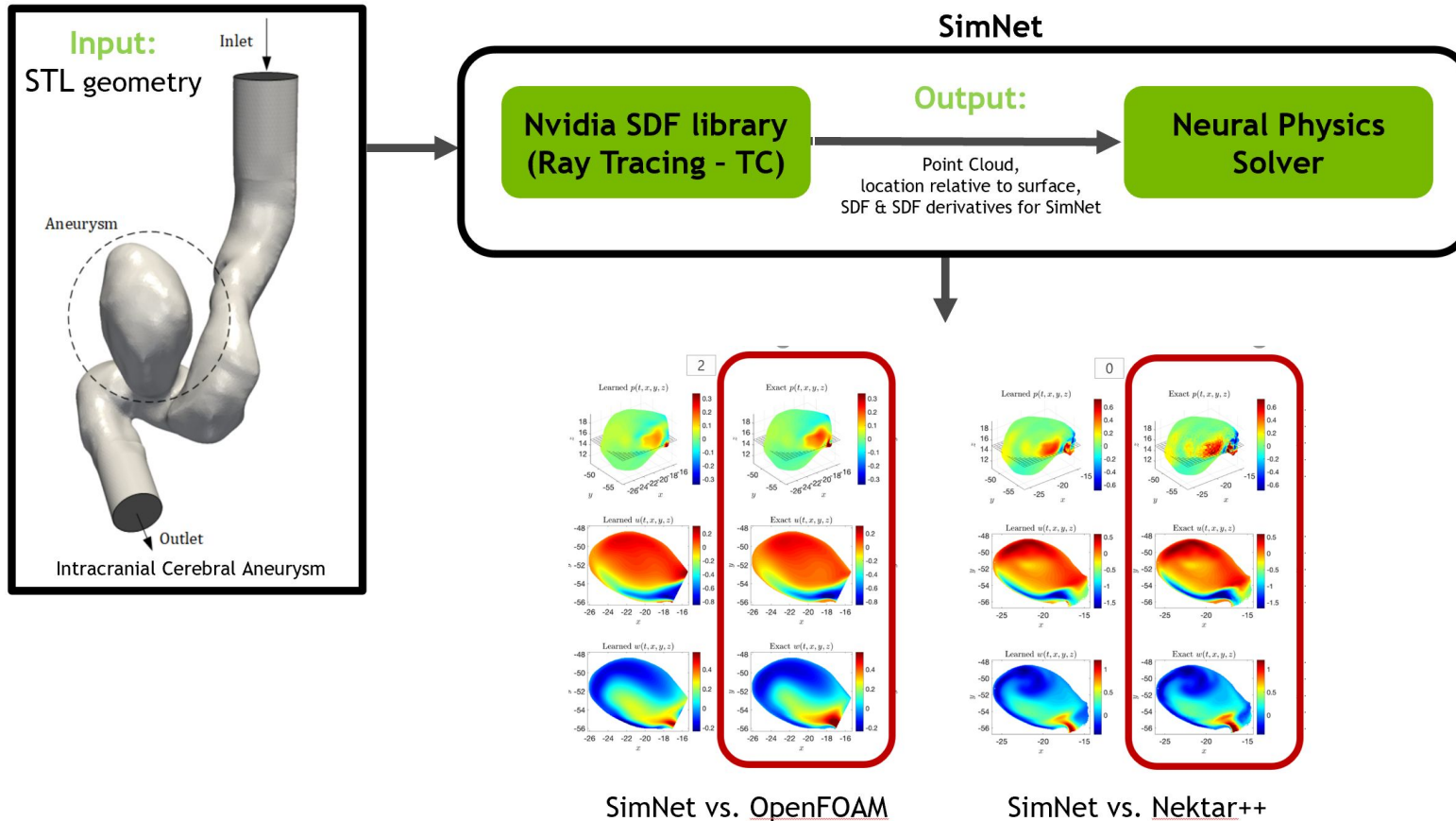
Results:

Property	OpenFOAM (True)	SimNet (Predicted)
Kinematic Viscosity (m^2/s)	1.00×10^{-2}	1.03×10^{-2}
Thermal Diffusivity (m^2/s)	2.00×10^{-3}	2.19×10^{-3}

TRANSIENT: MEDICAL IMAGING OF AN ICA

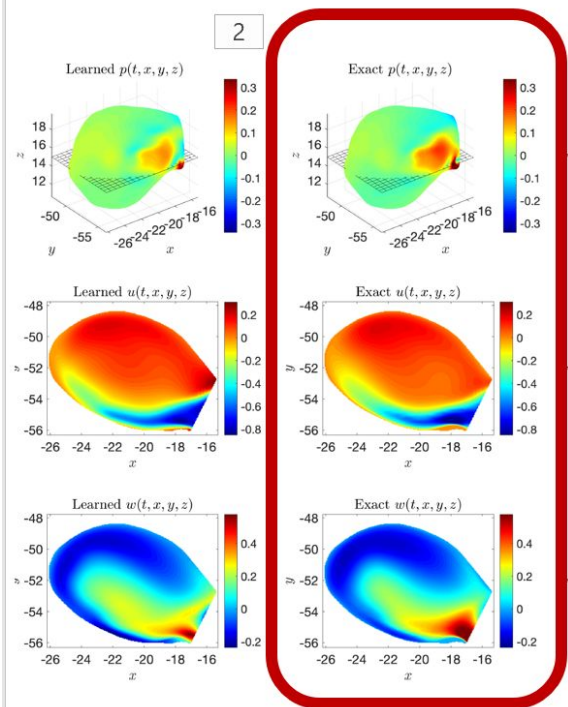
A Data Assimilation Problem

https://www.youtube.com/watch?v=QjY_8xFjsgE

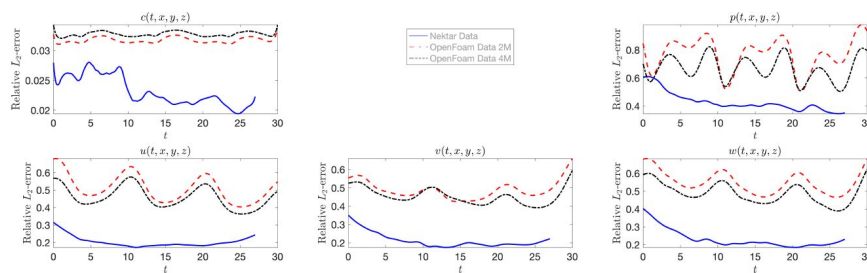


ICA – COMPARISON BETWEEN SIMNET & CFD SOLVERS

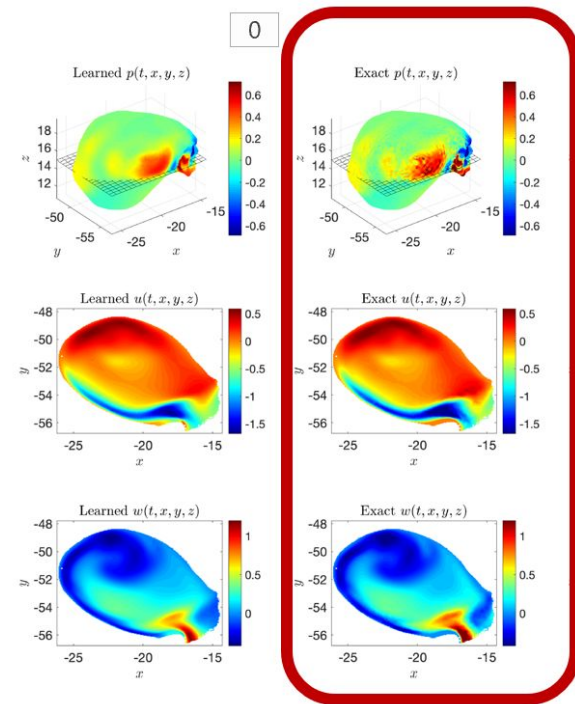
Inverse Solution



OpenFOAM vs.
Neural Networks



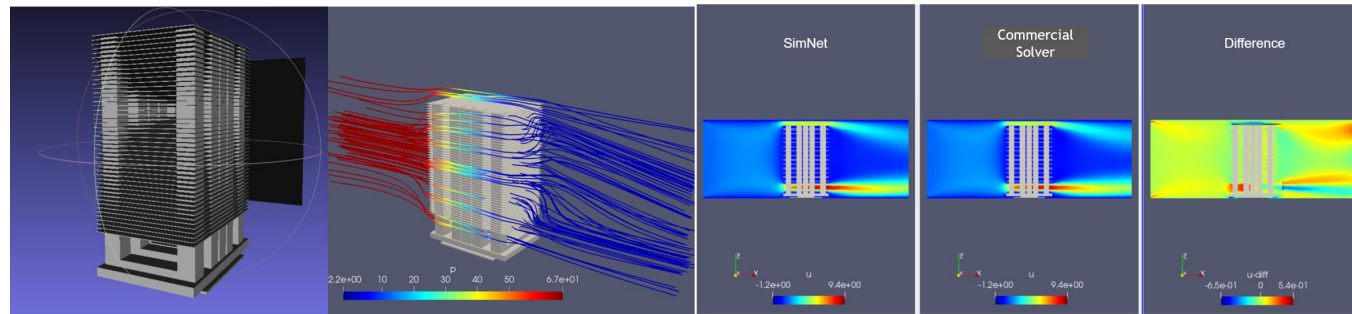
Nektar++ is a higher fidelity solver
(implicit, h- & p- method based
finite element CFD code) and
provides higher quality results with
less diffusion



Nektar++ vs.
Neural Networks

NVIDIA DGX-A100 NVSWITCH HEAT SINK

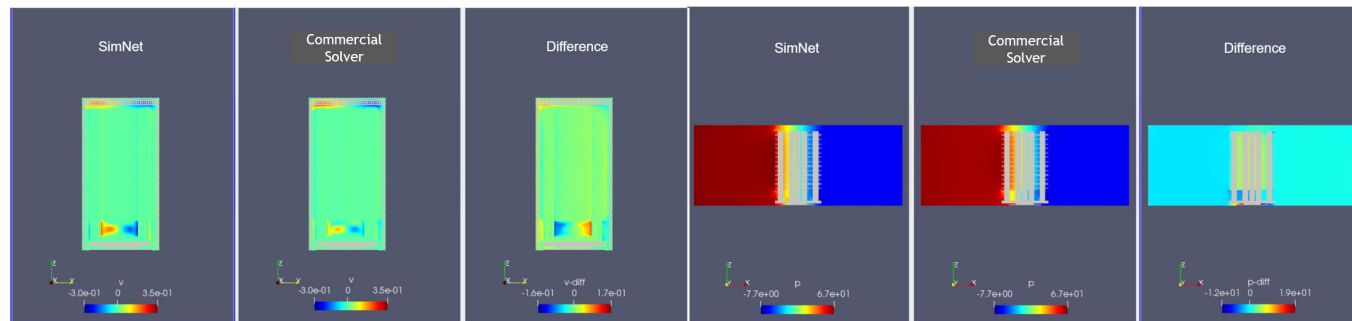
Multi-Physics Application: Fluids + Heat Transfer



Nvidia DGX A100 Heatsink

Pressure color coded flow streamlines

U-velocity (transverse direction) comparison



V-velocity (transverse direction) comparison

Pressure comparison

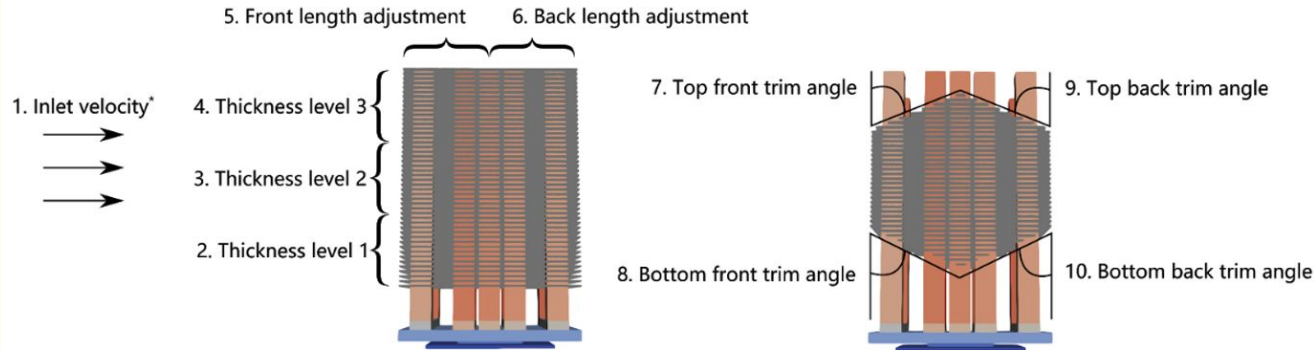
Turbulent Flow (Re=20,687)

	Temperature	Pressure Drop
SimNet - Fourier Network	43.1 °C	3.56
Commercial Solver	43.5 °C	3.6
OpenFOAM	41.6 °C	4.58

PARAMETERIZED DGX-A100 NVSWITCH HEAT SINK

Multi-Physics Application: Fluids + Heat Transfer

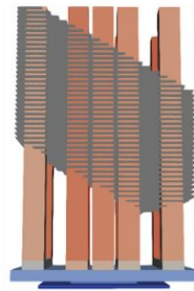
Limerock Design Variables



*Inlet velocity is not a design variable. It will be used for robust design optimization of Limerock in future.

Limerock Optimal Design

1. Inlet velocity: 5.7 m/s
2. Thickness level 1: 0.0031
3. Thickness level 2: 0.0044
4. Thickness level 3: 0.0030
5. Front length adjustment: -0.0124
6. Back length adjustment: 0.0025
7. Top trim angle: 0.0223 rad
8. Bottom front trim angle: 0.5197 rad
9. Top back trim angle: 0.5147 rad
10. Bottom back trim angle: 0.2217 rad



Max allowed pressure drop: 2.59
Number of random design evaluations: 4M

Peak temperature: 38.25 degC
Pressure drop: 2.5896

Computational Times (10 parameters, 3 values per parameter)

SimNet

1000 V100 GPU hrs.

Traditional Solver (OpenFOAM)
59,049 separate runs
(26 wall hours on 12 CPU cores)

18.4 Million CPU hrs.

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